

SUPPLEMENTAL TYPE CERTIFICATE

10063579

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AEROSPACE LOGIC INC.

**684 MUD STREET EAST, HANGAR 7
STONEY CREEK ON L8J 3C9
CANADA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: See "200 Series Eligibility List", Document # S200-CEL, Rev. 1.2, dated 21.12.2009 or later TCCA approved revisions.

Type Certificate Holder: See "200 Series Eligibility List".

Type: See "200 Series Eligibility List".

Model: See "200 Series Eligibility List".

Original STC Number: TCCA STC SA09-96

Description of Design Change:

Aerospace Logic Aircraft Instruments 200 Series as per Document # S200-CPL Rev 1.0 dated 02.12.2009 or later Transport Canada approved revisions.

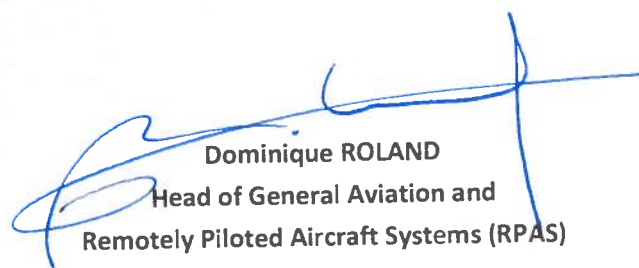
EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 25 October 2017



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)



The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

- Installation as primary or secondary, new or replacement must be in accordance with Aerospace Logic Certified Installation List – Document # S200-CIL, Rev. 1.0 dated 02/12/2009 (approved Transport Canada on 14/12/2009)
- Operation of instruments utilizing and intensity control module must be done in accordance with Aerospace Logic Document # S200-FMS, Rev. 1.0 dated 02/12/2009 (approved Transport Canada on 14/12/2009)
- Installation limitations must be observed in accordance with Aerospace Logic Certified Limitations Statement – Document # S200-ILS, Rev. 1.0 dated 02/12/2009 (approved Transport Canada on 14/12/2009)
- Maintenance must be in accordance with Aerospace Logic Instructions for Continued Airworthiness – Document # S200-ICA, Rev. 1.0 dated 02/12/2009 (approved Transport Canada on 14/12/2009) or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -